

Radar and Near-Earth Asteroid Exploration Missions

Michael Busch

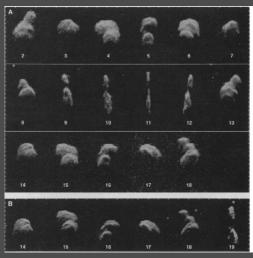
SSERVI Exploration Science Forum

2014 July 21

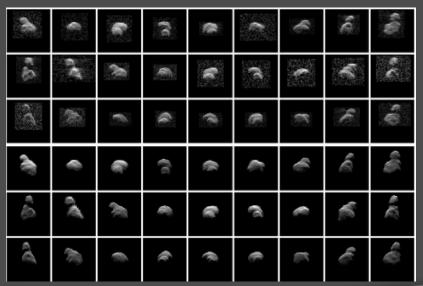
Case Study #1: Toutatis

Toutatis in 1992-1996

Work led by Steve Ostro and Scott Hudson

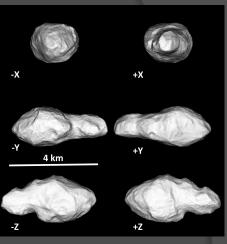


1992 Arecibo & Goldstone Radar Images

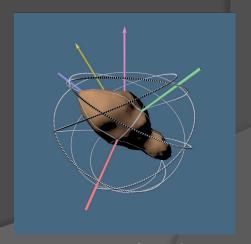


1996 Arecibo & Goldstone Radar Images



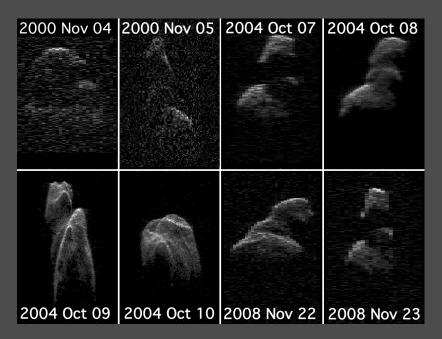


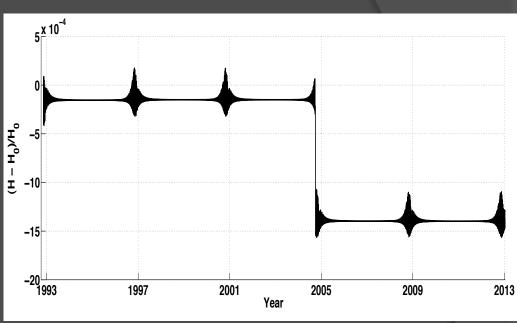
<u>Old</u> Radar-derived shape of Toutatis.



Toutatis' NPA Spin State

Toutatis in 2000-2008





Changes in Toutatis' angular momentum from Dec 1992 to Dec 2012. Chart and fit from Takahashi, Busch, & Scheeres 2013.

- Radar images: Goldstone '00, Arecibo '04 & '08.
- Mismatch between '92-'96 spin state fit and later images.
- Toutatis is torqued by gravitational tides from other objects. The largest spin state change since 1992 was during the '04 flyby.

Spin State Model

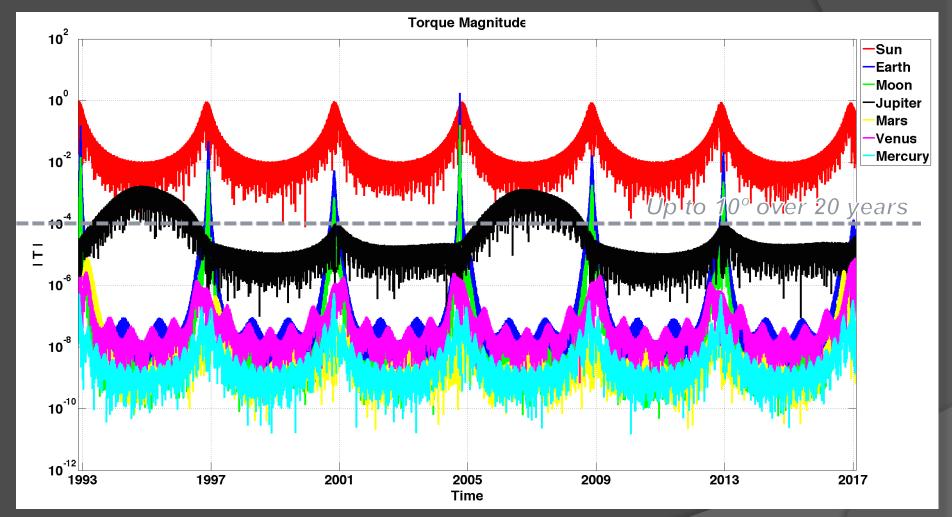
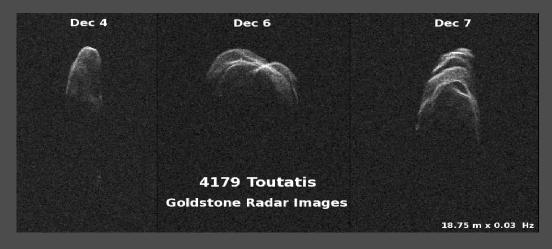
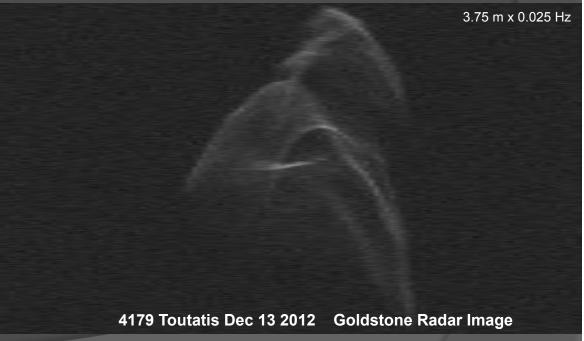


Chart by Yu Takahashi

Takahashi, Busch, Scheeres (2013) included torques from Earth and Sun. No longer good enough. Now adding Moon and Jupiter.

Toutatis in 2012





Toutatis Seen By Radar and Spacecraft



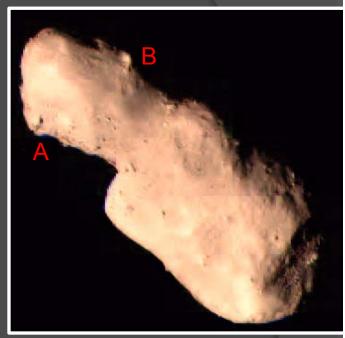


Image from Chang'e 2 (Huang et al. 2013)

- Radar-derived spin state model is quite good.
- Old radar shape model limited by viewing geometry and fitting procedure.
- Radar and spacecraft see many of the same surface features.
- Radar astrometry on Toutatis gave the asteroid's position to ± 600 m (3 σ).

Case Study #2: 2008 EV5

2008 Radar Campaign



Arecibo



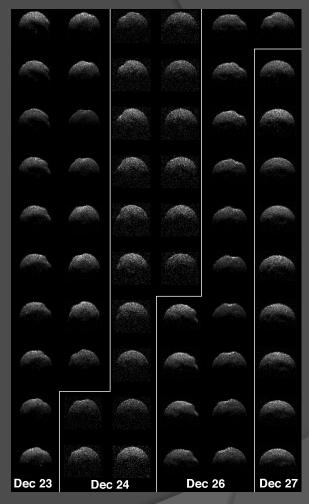
Goldstone 70-m



Green Bank



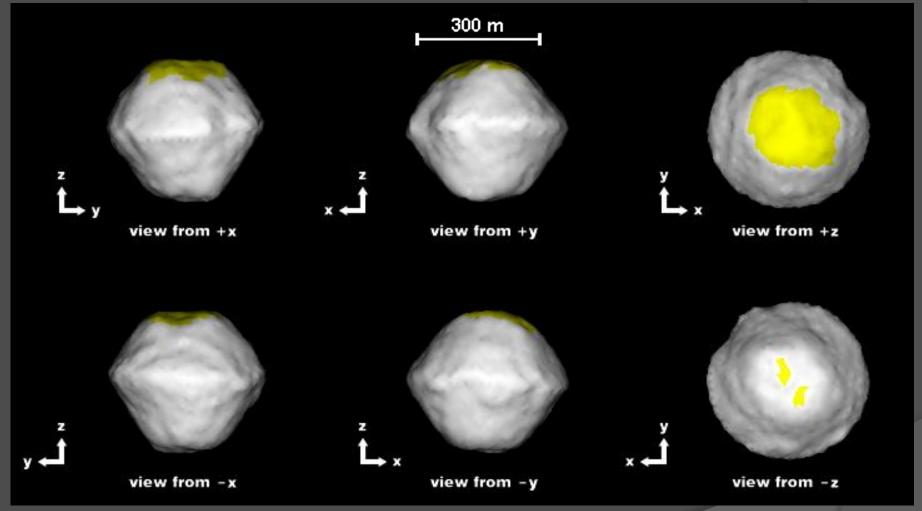
Concavity Concavity DEC 26 05:58 240° DEC 26 06:33 295° Concavity DEC 26 07:08 350° DEC 27 06:27 115°



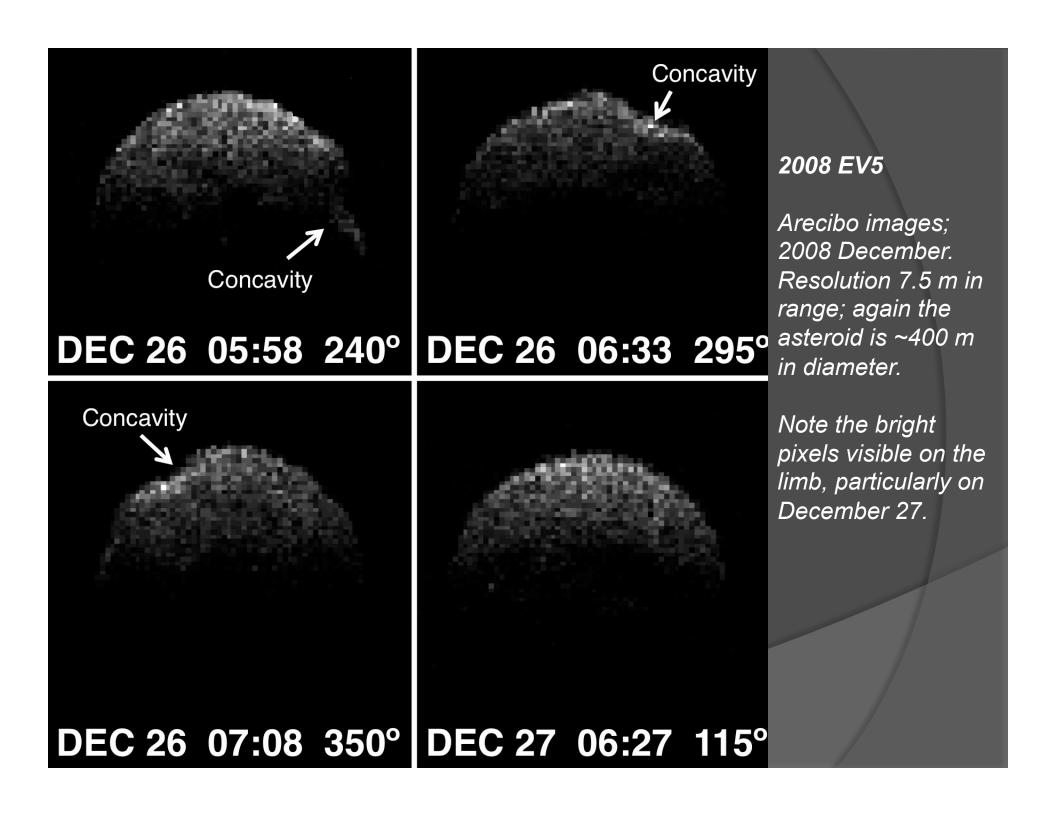
Arecibo delay-Dopper images of EV5; 7.5 m/pixel range resolution

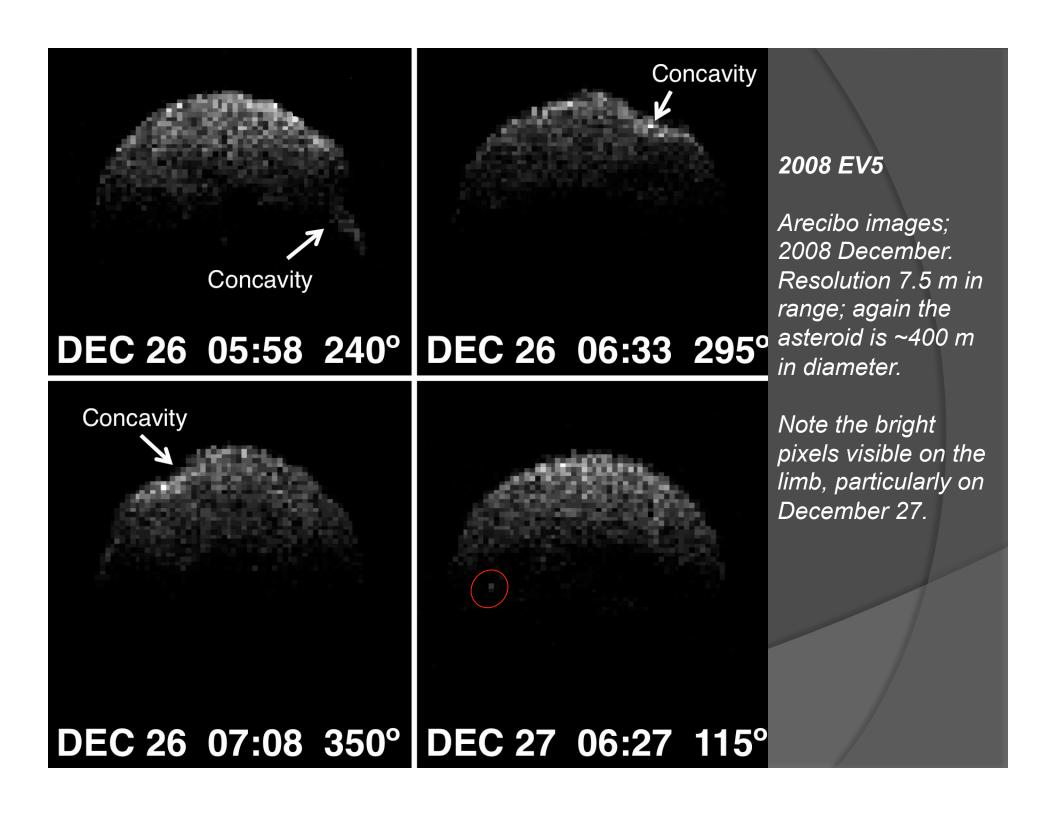
Very Long Baseline Array

EV5's Size And Shape

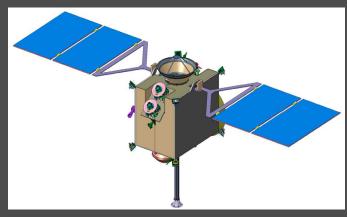


- Best-Fit Shape: ~400 m spheroid. The ridge is on the equator, and the concavity breaks the ridgeline.
 - Caveat: The poles, particular the north pole, were not seen or were seen only at grazing incidence.

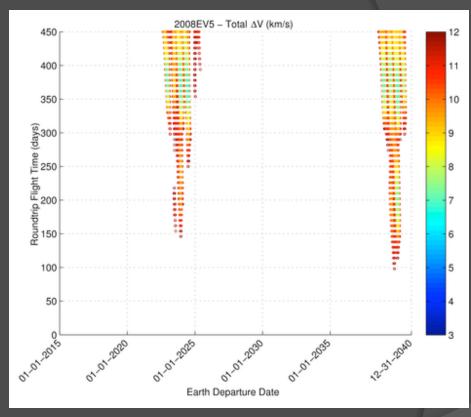




Proposed Missions to EV5 - I



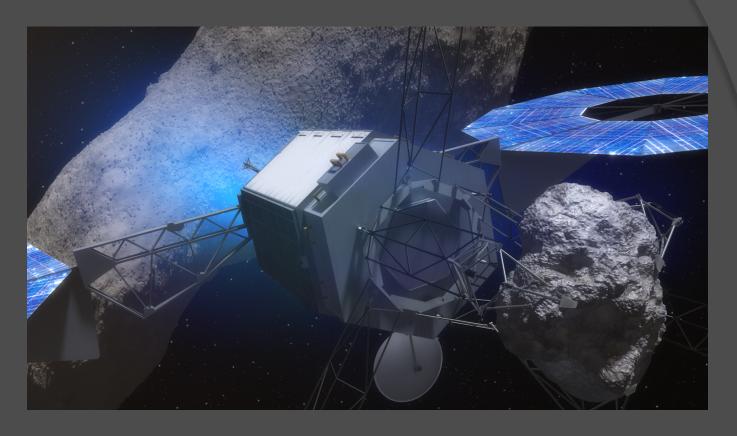
Marco Polo R. Proposed launch 2022-2024. In February 2014, ESA elected to not pursue the mission



NHATS. Human mission trajectories to EV5 in 2024 and 2039. 6.25 – 7 km/s Δv from LEO, ~1 year round-trip.

May be excluded as human NEO mission target due to long mission duration.

Proposed Missions to EV5 - II



ARM / ARRM Pick-up-a-rock. Would return 30-45 t boulder from EV5 to Earth-Moon space.

Launch c. 2019-2022; return c. 2025-2027.

NASA decision pending